

This examination has 5 questions on 5 pages.

The University of British Columbia  
Final Examinations—April 2010

Mathematics 403

Stabilization and Optimal Control of Dynamical Systems (Professor Loewen)

Open book examination.

Time:  $2\frac{1}{2}$  hours

Any resources used in class may be used during the examination.

Write your answers in the official examination booklet. Start each solution on a separate page.

[20] 1. We explore two feedback strategies,  $U_1$  and  $U_2$ , for the frictionless pendulum with forcing:

$$\dot{x}_1 = x_2, \quad \dot{x}_2 = -\sin(x_1) + u.$$

Note that the origin,  $(0, 0)$ , is an equilibrium point when  $u \equiv 0$ .

In parts (a)–(b), let  $U_1(x_1, x_2) = -\alpha x_2$  for some constant  $\alpha > 0$ .

- (a) Find all equilibrium points for the system when  $u = U_1(x_1, x_2)$ . Explain why this control cannot drive every trajectory to the origin.
- (b) Use a suitable Liapunov function to show that  $u = U_1$  makes the origin an asymptotically stable equilibrium point. Estimate the set of initial points  $(x_1(0), x_2(0))$  from which this feedback law will drive the trajectory to  $(0, 0)$  in the limit as  $t \rightarrow \infty$ .

In parts (c)–(d), let  $U_2(x_1, x_2) = \sin(x_1) - \alpha x_1 - \alpha x_2$  for some constant  $\alpha > 0$ .

- (c) Explain why using  $u = U_2$  makes the solution from every initial point in the plane converge to  $(0, 0)$  as  $t \rightarrow \infty$ .
- (d) Find a (quadratic) Liapunov function  $V$  that confirms the statement in (c).

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Useful Facts

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$$\begin{bmatrix} a & b \\ c & d \end{bmatrix}^{-1} = \frac{1}{ad - bc} \begin{bmatrix} d & -b \\ -c & a \end{bmatrix}$$

$$|pq| \leq \frac{1}{2}(p^2 + q^2), \quad \forall p, q \in \mathbb{R}$$

$$0 \leq x - \sin(x) \leq \frac{x^3}{6}, \quad \forall x > 0$$

$$\dot{x} = Ax + Bu \implies x(t) = e^{A(t-r)}x(r) + \int_r^t e^{A(t-s)}Bu(s) ds$$

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[20] **2.** Consider the following time-varying matrix involving a real constant  $m$ :

$$A(t) = \begin{bmatrix} -1 + m \cos^2 t & 1 - m \sin t \cos t \\ -1 - m \sin t \cos t & -1 + m \sin^2 t \end{bmatrix}.$$

- (a) Show that the eigenvalues of  $A(t)$  are independent of  $t$ , and identify the set of  $m$ -values for which both eigenvalues have negative real parts.
- (b) Show that the time-varying linear system  $\dot{\mathbf{x}} = A(t)\mathbf{x}$  has a solution of the form below, where  $k$  is some constant defined in terms of  $m$ :

$$\mathbf{x}(t) = e^{kt} \begin{bmatrix} \cos t \\ -\sin t \end{bmatrix}.$$

Identify the set of  $m$ -values for which  $|\mathbf{x}(t)| \rightarrow +\infty$  as  $t \rightarrow \infty$ .

- (c) Label each statement below as True or False, and give reasons. Assume all time-varying functions are smooth.
  - (i) For every constant matrix  $A$  in which each eigenvalue has a negative real part, every solution  $\mathbf{x}$  of  $\dot{\mathbf{x}} = A\mathbf{x}$  obeys  $\mathbf{x}(t) \rightarrow \mathbf{0}$  as  $t \rightarrow \infty$ .
  - (ii) For every matrix-valued function  $A = A(t)$  such that each eigenvalue has a negative real part at each  $t$ , every solution  $\mathbf{x}$  of  $\dot{\mathbf{x}} = A(t)\mathbf{x}$  obeys  $\mathbf{x}(t) \rightarrow \mathbf{0}$  as  $t \rightarrow \infty$ .
  - (iii) For every scalar-valued function  $a = a(t)$  such that  $a(t) < 0$  for each  $t$ , every solution  $x$  of  $\dot{x} = a(t)x$  obeys  $x(t) \rightarrow 0$  as  $t \rightarrow \infty$ .

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[20] **3.** We propose to launch a rocket straight up. Rockets generate thrust by using a chemical reaction to force part of their initial mass downward, so the rocket state has three time-varying components:  $z$ , the vertical position;  $v$ , the vertical velocity; and  $m$ , the mass. Taking the acceleration of gravity to be the constant  $g$ , we arrive at the system

$$\frac{dz}{dt} = v, \quad \frac{dv}{dt} = \frac{u}{m} - g, \quad \frac{dm}{dt} = -ku. \quad (1)$$

At launch, the rocket has altitude 0 [m], velocity 0 [m/s], and mass  $m_0 + m_1$  [kg]:

$$z(0) = 0, \quad v(0) = 0, \quad m(0) = m_0 + m_1. \quad (2)$$

Here  $m_0$  is the mass of the rocket shell and  $m_1$  is the mass of the fuel at launch time. Given some time  $T > 0$ , we seek the function  $u$  that will maximize the altitude  $z(T)$  while respecting the total-fuel constraint. This leads to the problem below, in which the final time  $T > 0$  is fixed:

Maximize  $z(T)$

Subject to dynamics (1) with initial conditions (2),

$$u(t) \in [0, 1], \text{ a.e. } t \in [0, T],$$

$$\int_0^T u(t) dt = \gamma \stackrel{\text{def}}{=} \frac{m_1}{k}.$$

We assume that  $T > \gamma$ , so that maximum thrust cannot be sustained for the full interval of interest.

- (a) Convert to a minimization problem and write the conditions describing an extremal control strategy,  $\hat{u}$ .
- (b) Show that the problem has no abnormal extremals.
- (c) Find the unique extremal control strategy and the corresponding value of  $z(T)$ .  
[Hint:  $\hat{u}(t)$  maximizes over  $u \in [0, 1]$  some function of the form  $s(t)u$ . Investigate  $\dot{s}(t)$ .]

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[20] 4. In this problem  $\varepsilon > 0$  is a given constant. Given a starting point  $(x_0, v_0)$ , the object is to choose  $T > 0$  and a function  $u: [0, T] \rightarrow \mathbb{R}$  so as to

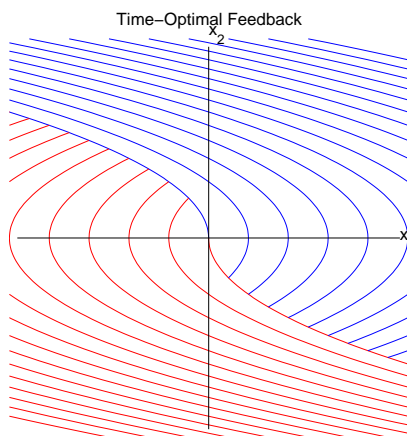
$$\begin{aligned} &\text{Minimize } \int_0^T (1 + \varepsilon|u(t)|) dt \\ &\text{Subject to } \dot{x}_1 = x_2 \quad \text{a.e.}, \quad x_1(0) = x_0, \quad x_1(T) = 0, \\ &\quad \quad \dot{x}_2 = u \quad \text{a.e.}, \quad x_2(0) = v_0, \quad x_2(T) = 0, \\ &\quad \quad u(t) \in [-1, 1] \quad \text{a.e.} \end{aligned}$$

(When  $\varepsilon = 0$ , the integral simplifies to  $T$ , and the problem reduces to the familiar task of steering the rocket car to the origin of phase space in minimum time. But now  $\varepsilon > 0$ , and its role is to discourage excessive use of control input.)

(a) Derive the following general fact, which will be useful later:

$$\arg \max_{w \in [-1, 1]} \{rw - |w|\} = \begin{cases} -1, & \text{if } r < -1, \\ 0, & \text{if } -1 < r < 1, \\ 1, & \text{if } r > 1. \end{cases}$$

- (b) Some extremal control functions have the value  $+1$  on a nondegenerate final interval of the system path. Use the Pontryagin Maximum Principle to describe the most general form of an extremal control with this property. (When  $\varepsilon = 0$ , this was an important observation in the rocket car solution: either the control was  $+1$  forever, or else the final interval was preceded by exactly one interval where the control was  $-1$ .)
- (c) Construct an optimal feedback strategy for the portion of phase space in which extremals of the type in (b) provide the evolution. The desired result is a replacement for the upper part of this familiar picture of the phase space for the problem with  $\varepsilon = 0$ :



Specify the equations for any switching curves you find.

*Hint:* Remember that the preHamiltonian  $\hat{h}(t) \stackrel{\text{def}}{=} H(t, \hat{x}(t), p(t), \hat{u}(t))$  is constant along every extremal, with a value determined by the transversality condition appropriate for problems on variable time intervals.

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[20] **5.** Consider the forced harmonic oscillator on the interval  $[\tau, 2\pi]$  with a free final state. Given  $k_1 \geq 0$  and  $k_2 \geq 0$  (not both 0), the problem is

$$\text{minimize } J[u] \stackrel{\text{def}}{=} \frac{k_1}{2} (y(2\pi))^2 + \frac{k_2}{2} (\dot{y}(2\pi))^2 + \int_{\tau}^{2\pi} \frac{1}{2} |u(t)|^2 dt,$$

over all  $u: [\tau, 2\pi] \rightarrow \mathbb{R}$  piecewise continuous

subject to  $\ddot{y}(t) + y(t) = u(t)$ , a.e.  $t \in [\tau, T]$ ,

$$y(\tau) = \xi_1, \dot{y}(\tau) = \xi_2.$$

- (a) Identify extremal strategies for this problem.
- (b) Write the true Hamiltonian for this problem.
- (c) If  $V(\tau, \xi)$  denotes the minimum value in the stated problem [and  $V$  happens to be continuously differentiable], then it must satisfy a certain first-order partial differential equation (PDE) and boundary condition (BC). Write these down.
- (d) In order for the PDE/BC in part (c) to have a solution of the form

$$V(\tau, \xi) = \frac{1}{2}a(\tau)\xi_1^2 + b(\tau)\xi_1\xi_2 + \frac{1}{2}c(\tau)\xi_2^2,$$

what conditions must be satisfied by the functions  $a$ ,  $b$ , and  $c$ ?

- (e) The answer in (d) involves a system of ordinary differential equations for the unknown functions  $a$ ,  $b$ , and  $c$ . Find as many equilibrium points for this system as you can.